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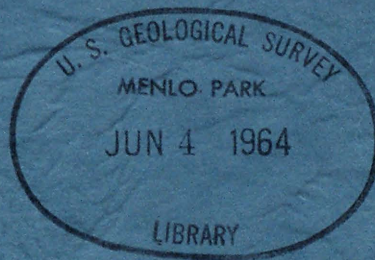
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Solution of the Gamma-Ray Transport
Equation for Two Media, the Ground
Source and Air, in Airborne Radio-
activity Surveying.

by
Sakakura, A. Y.



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by A. Y. SAKAKURA

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standards or nomenclature.



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SOLUTION OF THE GAMMA-RAY TRANSPORT EQUATION FOR TWO MEDIA,
THE GROUND SOURCE AND AIR, IN AIRBORNE RADIOACTIVITY SURVEYING.

INTRODUCTION

As the gamma-ray flux measured in airborne radioactivity surveying comes predominantly from a ground source essentially both infinitely thick and extensive for the radiation in question, a detailed solution of the steady state transport equation for gamma rays would be both interesting and useful. Such a solution for the two media, the ground source and air, would provide both the angular and spectral distribution of the gamma-ray flux from the uranium, thorium, and potassium series, and would establish the true total flux in air at various distances from the source.

Two methods of solution are available: the moments method of Spencer and Fano (1951), and the direct integration method of Keller and Heller (1955). The first method has been used with great success, and extensive series of calculations have been performed for one medium by Goldstein and Wilkins (1954). The second method, the direct numerical integration of the steady state transport equation for neutrons with elastic scattering and an isotropic source (linearized Boltzmann), has also been successfully tested; this method shows that "small" errors introduced at the boundaries influence the solution only by a "small" amount, and that by taking a sufficiently fine mesh the numerical solution can be made as close as desired to the true transport solution.

1. The angular distribution of the flux in air at 500 feet from the air-source interface is approximately isotropic, and two or three Legendre moments are sufficient to characterize the angular flux. In the higher energy region, removal of the once-collided term as was done by Certaine (1953) would insure the need of but a few moments. 2. The angular distribution of the flux in air at 500 feet from the air-source interface is approximately isotropic, and two or three Legendre moments are sufficient to characterize the angular flux. In the higher energy region, removal of the once-collided term as was done by Certaine (1953) would insure the need of but a few moments. 3. The neutron problem had already been programmed for the Univac computer, and modification of the neutron code necessary for the gamma-ray problem would be simple.

The first reason is more apparent than real as in the range where the Compton scatter effect dominates, the two-media problem can be cast into an equivalent one-medium problem. Although this is not possible in the low energy region, the flux in this region would be quite sensitive to the nature of the source material; consequently, one might as well proceed with an equivalent one-medium problem. After computing several problems within the overall problem, it became apparent that the second reason was no longer valid. In the low energy region, the flux in air at 500 feet from the air-source interface is approximately isotropic, and two or three Legendre moments are sufficient to characterize the angular flux. In the higher energy region, removal of the once-collided term as was done by Certaine (1953) would insure the need of but a few moments.

A total of 11 problems were programmed and computed under the code designation, Ariel, at the AEC Computing Facility at New York University. These problems correspond to the primary spectrum of RaC, Th C", and K⁴⁰ at the initial energies of 2.42, 2.63, and 1.45 mev respectively. The primary RaC spectrum was that given by Mladjenovic and Hedgran (1954); the thorium spectrum was that determined by N. Lazar of the Oak Ridge National Laboratory.

The RaC and thorium spectrum were normalized to 1 primary gamma source. The cross-sections were taken from White (1952) with constant cross-sections included.

The ground source was taken to be silicon dioxide of density 2.65 g/cm³. The interface between air and the ground source was taken to be sea-level. It should be noted that for ground sources of different densities, the solution will apply if the concentration of gamma-emitting isotopes were expressed in terms of disintegrations per gram. The change of the air-ground interface from sea level to any higher altitude would require adjusting distance from source to the equivalent air density at sea level. Although the effect of the "law of atmosphere" is small compared to other errors, it was included in the computations inasmuch as it was simple to do so.

The raw output of the computer is the angular flux at various energies and distances from the air-source interface. However, the computer print-out data are not included in this report because of exceedingly large volume, equivalent to many hundreds of pages. These data are available in the files of the Geological Survey, and those who are interested may make copies.

ACKNOWLEDGMENTS

We are grateful to E. Bromberg, Director of the AEC Computing Facility, Institute of Mathematical Sciences, New York University, for the hospitality extended to Sakakura during his protracted stay at the Institute. We are particularly indebted to J. Heller of the Institute for modifying the Achilles code for integration of the neutron transport equation to the Ariel code for the gamma-ray transport equation, for his continued support in the debugging process, and for his overseeing the running of the problems on the computer. We thank N. Lemar of the Oak Ridge National Laboratory for determining the uranium and thorium spectra.

The work herein reported is part of an investigation carried on by the U. S. Geological Survey on behalf of the U. S. Atomic Energy Commission.

$\lambda = \lambda_0 / \gamma$, and $\lambda_0 = \lambda \gamma$, where $\gamma = 1 / \sqrt{1 - \beta^2}$.

λ = wave length, in Compton units, related to unit λ_0 in air by

$$\lambda = \frac{0.511}{E}$$

$Y^i(x, \theta; \lambda_0)$ = angular flux, the number of photons in wave-length

interval, $(\lambda, \lambda + d\lambda)$, in a solid angle, $d\Omega$, crossing the unit area

normal to $\vec{u}$ per unit time due to a semi-infinite isotropic source

emitting 1 photon of wave length, λ_0 , per unit time per unit volume.

$e^{-x/\ell}$ = factor representing the "law of atmosphere"

$K_0(\lambda^1, \lambda)$ = linear differential Compton scattering cross-section from

λ^1 to λ for the source medium

$K_2(\lambda^1, \lambda)$ = linear differential Compton scattering from λ^1 to λ for

air at sea level density

$\mu_B(\lambda)$ = linear total attenuation cross-section for air at sea level density

photon of wave length, λ .

$\mu_0(\lambda)$ = linear total attenuation cross-section for air at sea level

density

$[\lambda_0, \lambda - 2\lambda^1]$ = greater of the two λ 's.

geometry is depicted in Figure 1. To remove the delta (δ)

function singularity, we write Y^1 as the sum of the unscattered flux and the scattered flux, Y :

$$Y^1(x, \lambda, z; \lambda_0) = Y(x, \lambda, z; \lambda_0) + \frac{\lambda(\lambda - \lambda_0)}{4\pi\mu_s(\lambda_0)} r(x, z) \quad (2)$$

where

$$\begin{aligned} r(x, z) &= 1 - e^{-\frac{\mu_s(\lambda_0)x}{z}} \text{st}(-z), \quad x < 0 \\ &= \exp\left\{\frac{-\mu_s(\lambda_0)\lambda(1 - e^{-x/\lambda})}{z}\right\} \text{st}(z), \quad x > 0 \end{aligned} \quad (3)$$

Y , then, obeys the same transport equation (1) as Y^1 except that there is now a source term, S , in both media:

$$\begin{aligned} S_2(x, \lambda, z; \lambda_0) &= \frac{K_2(\lambda_0, \lambda)e^{-x/\lambda}}{4\pi\mu_s(\lambda_0)} \int_{\lambda_0}^{\lambda} \frac{du}{u} \frac{u^2}{u^2 - 1 - \lambda_0 + \lambda} r(x, z^1; \lambda_0), \quad x > 0, \lambda - \lambda_0 < 2 \\ S_1(x, \lambda, z; \lambda_0) &= \frac{K_1(\lambda_0, \lambda)}{4\pi\mu_s(\lambda_0)} \int_{\lambda_0}^{\lambda} \frac{du}{u} \frac{u^2}{u^2 - 1 - \lambda_0 + \lambda} r(x, z^1; \lambda_0), \quad x < 0, \lambda - \lambda_0 < 2 \\ &= 0, \text{ all } x \text{ and } -\lambda_0 > 1 \end{aligned} \quad (4)$$

Now let

$$\begin{aligned} \phi_2(x, \lambda, z) &= 0, \quad x < 0 \\ &= (1 - e^{-x/\lambda}), \quad x > 0 \end{aligned}$$

then

$$K(x, \lambda, z; \lambda_0) = \int_{[\lambda_0, \lambda-2]}^{\lambda} K_s(\lambda^1, \lambda) d\lambda^1 \int_{4\pi} d\vec{u} \frac{\delta(\vec{u} \cdot \vec{u} - 1 - \lambda^1 + \lambda)}{2\pi} Y(x, \lambda^1, z^1; \lambda_0), x < 0$$

$$= \int_{[\lambda_0, \lambda-2]}^{\lambda} K_s(\lambda^1, \lambda) d\lambda^1 \int_{4\pi} d\vec{u} \frac{\delta(\vec{u} \cdot \vec{u} - 1 - \lambda^1 + \lambda)}{2\pi} Y(x, \lambda^1, z^1; \lambda_0), x > 0$$

and define

$$S(y, \lambda, z; \lambda_0) = S_a(x, \lambda, z; \lambda_0), x < 0$$

$$= S_b(x, \lambda, z; \lambda_0), x > 0$$

We change to discrete values of the wave-length, λ , and perform the

integration by the trapezoidal rule, resulting in

$$z \frac{\partial Y}{\partial y}(x, \lambda_n, z; \lambda_0) + \sigma(\lambda_n) Y(x, \lambda_n, z; \lambda_0) = K^1(x, \lambda_n, z; \lambda_0) + S(x, \lambda_n, z; \lambda_0) \quad (5)$$

where

$$K^1(x, \lambda_n, z; \lambda_0) = K(x, \lambda_n, z; \lambda_0) - K(x, \lambda_n, \lambda_n) \frac{(\lambda_n - \lambda_{n-1}) Y(x, \lambda_n, z; \lambda_0)}{2}$$

$$\sigma(\lambda_n) = \mu(\lambda_n) - K(\lambda_n, \lambda_n) \frac{(\lambda_n - \lambda_{n-1})}{2} = \mu(\lambda_n) - \frac{3}{2} \sigma_t (\lambda_n - \lambda_{n-1})$$

$$\sigma_t = \text{Thompson cross-section}$$

The right hand side no longer contains $Y(x, \lambda_n, z; \lambda_0)$, the unknown, and can

thus be integrated:

$$Y(x \pm \delta x, \lambda_n, z; \lambda_0) = Y(x, \lambda_n, z; \lambda_0) \exp \left[\frac{-\delta y \sigma(\lambda_n)}{z} \right] \quad (6)$$

$$+ \frac{1}{z} \int_y^{y+\delta y} \exp \left[\frac{-\sigma(\lambda_n)(y+\delta y-y')}{z} \right] \left\{ K^1(x^1, \lambda_n, z; \lambda_0) + S(x^1, \lambda_n, z; \lambda_0) \right\} dy^1, z \gtrless 0$$

where $\delta y(x) = y(x \pm \delta x) - y(x)$

[illegible]

Figure 1 is a schematic diagram of the experimental setup. It shows a subject sitting at a table, viewing a video screen. A horizontal line is projected onto the screen. A vertical line is also shown, intersecting the horizontal one. The subject's eye is positioned to observe the intersection point. Labels include 'Subject', 'Video Screen', 'Horizontal Line', 'Vertical Line', and 'Intersection Point'.

In order to integrate the collision term, $\Pi(x, y, z; \lambda)$ over y , it is necessary to use the discrete variable, x_k , rather than the continuous, x . We define, $x_{k+1} = x_k + \Delta x$ and $y_k = y(x_k)$, and interpolate:

$$K^2(x_k^2, \lambda_n, z; \lambda_0) \sim -K^2(x_k, \lambda_n, z; \lambda_0) \frac{(y - y_k - \delta y_k)}{\delta y_k} + K^2(x_k + \delta x, \lambda_n, z; \lambda_0) \frac{(y - y_k)}{\delta y_k} \quad (9)$$

The integration in (10) can be performed:

$$\frac{1}{z} \int_{-\infty}^{\infty} \frac{e^{-\frac{1}{2}(\frac{x}{\lambda_N})^2 - \frac{1}{2}(\frac{y}{\lambda_N})^2}}{z} A(x', y', z; \lambda_0) \quad (12)$$

$$\frac{1}{\sigma(\lambda_N)} \left[-\frac{e^{-\frac{1}{2}(\frac{x}{\lambda_N})^2}}{z} + \frac{e^{-\frac{1}{2}(\frac{y}{\lambda_N})^2}}{z} \right] K^2(x_N, y_N, z; \lambda_0)$$

$$+ \frac{1}{\sigma(\lambda_N)} \left[1 - \frac{e^{-\frac{1}{2}(\frac{x}{\lambda_N})^2}}{z} \right] K^2(x_N = \infty, y_N, z; \lambda_0)$$

$$A(x_N, y_N, z; \lambda_0) K^2(x_N, y_N, z; \lambda_0) + B(x_N, y_N, z; \lambda_0) K^2(x_N, y_N, z; \lambda_0)$$

The transport equation thus takes the form

$$Y(x_N, y_N, z; \lambda_0) \quad (13)$$

$$= B(x_N, y_N, z; \lambda_0) \frac{e^{-\frac{1}{2}(\frac{x}{\lambda_N})^2}}{z} + A(x_N, y_N, z; \lambda_0) K^2(x_N, y_N, z; \lambda_0)$$

$$+ B(x_N, y_N, z; \lambda_0) K^2(x_N = \infty, y_N, z; \lambda_0) + T(x_N, y_N, z; \lambda_0)$$

to which Hammer's method (Keller and Hammer, 1959) can now be readily applied, if the present infinite boundaries can be replaced by finite ones. This has been accomplished by assuming that the air is bounded on the right at 1000 feet by vacuum which makes the incoming flux zero at that point, and that the left hand boundary within the source is established by

increasing and the flux is increasing there rather than the incoming flux decreasing there. Now, Miller's method can be directly applied, starting at the right hand boundary with increasing flux, turning at the left hand boundary with decreasing flux, and back to the right hand boundary.

Finally, in order to carry out the angular integration, we take J discrete values of z . Consider the integral

$$\int_{\lambda_0}^{\lambda_n} d\bar{u} \frac{c(\bar{u}, \bar{u}^2 - 1 - \lambda_0^2 + \lambda_n^2)}{4\pi} L(z^2) = I(z, \lambda_n, \lambda_0) \quad (12)$$

where $L(z^2)$ is a function of z^2 . (In $T(x, \lambda_n, z; \lambda_0)$, $L(z^2) = g(x, z^2, z; \lambda_0)$, $n < n$.) The integration over the azimuthal angle can be readily performed:

$$I(z, \lambda_n, \lambda_0) = \frac{1}{\pi} \int_0^\pi \frac{L(z^2) d\bar{u}}{\sqrt{(\beta - z^2)(z^2 - \alpha)}} \quad (13)$$

where $\alpha = 1 + \lambda_0^2 - \lambda_n^2$

$$\beta = 1 + \lambda_0^2 + \lambda_n^2$$

$$\gamma = 1 + \lambda_0^2 - \lambda_n^2$$

function, $st(z)$, is a linear combination of

$$L_1(z^1) + L_2(z^1) st(z^1 - z_m),$$

where L_1 and L_2 are well-behaved functions of z^1 , and st is a step function.

(For example, in the source term we can write $st(z^1 - z_m) = 1 - st(z_m - z^1)$. Here

$st(z_m - z^1) = 0$.) Integration is performed by the usual method. The integral

involving $L_1(z^1)$ is readily performed:

$$\frac{1}{\pi} \int_{\alpha}^{\beta} \frac{L_1(z^1)}{\sqrt{(\beta - z^1)(z^1 - \alpha)}} dz^1 = \frac{1}{J} \sum_{p=0}^{J/2} L_1(z_p^1), \quad z_p^1 = \frac{\beta - \alpha}{2} O_p + \frac{\beta + \alpha}{2} \quad (14)$$

where O_p 's are the roots of Tschobyscheff polynomial of degree $J/2$.

The integral involving the step function is:

$$\begin{aligned} \frac{1}{\pi} \int_{\alpha}^{\beta} \frac{L_2(z^1) st(z^1 - z_m)}{\sqrt{(\beta - z^1)(z^1 - \alpha)}} dz^1 &= 0 & \beta < z_m & \quad (15) \\ &= \frac{2}{J} \sum_{p=0}^{J/2} L_2(z_p^1) \sqrt{\frac{z_p^1 - z_m}{z_p^1 - \alpha}}, & \beta > z_m & > \alpha \\ z_p^1 &= \frac{\beta - z_m}{2} O_p + \frac{\beta + z_m}{2} \end{aligned}$$

and is identical to (14) for $\alpha \leq z_m$. For the source terms, the L 's can be

evaluated at z_p ; for the K^1 's, the Y 's are linearly interpolated between

two z 's bracketing z_p , i.e.

$$Y(x_k, \lambda_{n^1}, z_p) \sim Y(x_k, \lambda_{n^1}, z_q) \frac{(z_p - z_{q-1})}{\delta_z} + Y(x_k, \lambda_{n^1}, z_{q-1}) \frac{(z_q - z_p)}{\delta_z} \quad (16)$$

$$\text{where } \delta_z = \frac{2}{J-1}, \quad z_q = z_z + z_{q-1}$$

The energy E is divided into bins E_i which appear in K' is introduced by the Monte Carlo steps, which represent the number of collisions necessary to reach a given energy, and K' is a distribution on the values of E for the energies close to the initial energy. The use of K'_m insures that the numerical integration is restricted to the significant portion of the angular interval.

In the actual running of the program, the following parameters were used:

$$\Delta x_a = .05 \text{ ft.} \quad \Delta x_a = 100 \text{ ft.} \quad \Delta \theta_a = \frac{\pi}{4}$$

The thickness of air (to the right boundary) was taken to be 1000 feet; and the thickness of the source (to the left boundary) to be 5 mean free paths of the primary photon.

To determine the independence of the results to the choice in program 2.0, we have run a problem with the following parameters:

$$\Delta x_a = .05 \text{ ft.} \quad \Delta x_a = 50 \text{ ft.} \quad \Delta \theta_a = \frac{\pi}{4}$$

The thickness of the air was taken to be 1000 feet; and the thickness of

the source to be 7 mean free paths. Taking the initial energy to be 2.63 mev, the flux at 1.76 mev, which was separated from 2.63 mev by 5 intermediate energies instead of the usual two, was examined. As the flux within the source would be little influenced by the change of parameters in air, the differences in the two problems would show the effect of the change in δz , δx , and δt . Throughout the problem and at the boundary, the fluxes for the two problems differ by less than one percent, and we thus conclude that the δz , δx , and δt as determined are quite good for the source width. As the flux in air at $z = 1$ is lower relative to the value of δz and δx the boundary imposed on the inconsistency of the δz and δx as a function of the errors of change in δz . Again for the two problems, even at 100 feet the differences were less than one percent, indicating that $\delta x_0 = 100$ feet is sufficiently fine mesh.

Because of the sharply forward peaked nature of the high energy fluxes, the evaluation of the source terms depends strongly on the choice of δz when $z < 0$. The angular fluxes differ about five percent in the range $1 > z > 0.1$, about ten percent in the range $0.1 > z > 0$, and about an order of

unimportant. However, this error is not important as the flux in the backward direction is very small compared to the forward flux. Thus, the major contribution to the backward fluxes at energies close to the source energies ($E_1 \sim E_2$) is the scattering from the forward fluxes. At lower energies, the backward flux is of the same order of magnitude as the forward flux, and the major contribution is the scattering from higher energy backward fluxes which in turn come from scattering of high energy forward fluxes. Thus, the effect of source term inaccuracy and position of the right hand boundary on the backward flux is not serious at low energies whereas at high energies, the backward flux itself is not important. The accuracy of the Legendre moments is good, the difference being 17% percent at 1000 feet and two percent at 500 feet.

Discussion of general results

Following Goldstein and Wilkins (1954), we call $Y_0(x, \lambda)$ the flux, as contrasted to the angular flux $Y(x, \lambda, z)$, to which it is related by:

$$Y_0(x, \lambda) = \int_{4\pi} d\vec{\Omega} Y(x, \lambda, z) \quad (1)$$

The flux, divided by the velocity of light, is the total number of photons of wavelength λ , regardless of direction of travel, contained in a unit volume. We also introduce the spectral flux, $N(x, E)$ which is related to

$$\text{the flux by : } N(x, E) = - \frac{d\lambda}{dE} Y_0(x, \lambda) = \frac{0.0143}{\lambda^5} Y_0(x, \lambda) \quad (2)$$

which gives the total number of photons/unit area/sec. regardless of direction in the energy interval dE . They are related to the integral flux

$$Y_c(x, \lambda) = N_c(x, E) = \int_E^{E_0} N(x, E) dE = \int_{\lambda_0}^{\lambda} Y_0(x, \lambda') d\lambda' \quad (3)$$

The total integral flux is

$$Y_c^1(x, \lambda) = N_c^1(x, E) = N_c(x, E) + \text{primary.}$$

We also define the Legendre moments, $\psi_\ell(x, \lambda)$:

$$\psi_\ell(x, \lambda) = \frac{2\ell+1}{4\pi} \int_{-1}^{+1} Y_0(x, \lambda, z) P_\ell(z) d\vec{\Omega} \quad (4)$$

where $P_\ell(z)$ is Legendre polynomial of order ℓ .

Note that ψ_0 for a given source is Y_0 of a source $\frac{1}{4\pi}$ times as intense.

-1. In the case of the first two moments, the agreement is good. (1954) for the plane-wave directional source, agreement for the other energies is also good. The reason the fourth moment from plane waves is less than the value obtained for the plane-wave directional source is that the plane-wave directional source is more directional than the plane-wave source.

We finally compare (in table 5) the moments build-up curves for 2.43 to 0.4 mev, computed from Nuclear Development Associates Plumes (Goldstein and Wilkins, 1954) in a manner outlined in a previous paper (Sakakura, 1957), with the build-up computed from the plane waves. The results are in reasonable agreement. Note that the initial values for the moments are lower than the moments build-up curves, but they are in reasonable agreement with the moments build-up curves for the finite thickness of air.

In Figure 1 we have plotted the moments build-up curves for various energies to source energies of 2.63, 1.46, and 0.39 mev. Note that the moments build-up curves for the source of 2.63 and 1.46 mev are more isotropic at $E = 0.2$ mev. for the source of 2.63 and 1.46 mev energy, whereas the one due to the source of 0.39 mev. is much more anisotropic. Note also the fact that at $\lambda > \lambda_0 + 2$ the curves are roughly equal regardless of the initial energy due to the photons for attaining their initial energy after many collisions. At $E = 0.39$ mev, for which $\lambda < \lambda_0 + 2$ for all the

Figure 1 shows the angular distribution of the source energy in the forward direction.

$$\frac{I(\theta)}{I(0)} = \frac{1}{2} \left(1 + \cos \theta \right)$$

where $I(\theta)$ is the intensity at an angle θ from the forward direction.

The angular distribution of the source energy in the backward direction is shown in Figure 2.

Figure 3 shows the angular distribution of the source energy in the forward direction.

The angular distribution of the source energy in the backward direction is shown in Figure 4.

Figure 5 shows the angular distribution of the source energy in the forward direction.

Figure 6 shows the angular distribution of the source energy in the backward direction.

Figure 7 shows the angular distribution of the source energy in the forward direction.

Figure 8 shows the angular distribution of the source energy in the backward direction.

Figure 9 shows the angular distribution of the source energy in the forward direction.

Figure 10 shows the angular distribution of the source energy in the backward direction.

Figure 11 shows the angular distribution of the source energy in the forward direction.

Figure 12 shows the angular distribution of the source energy in the backward direction.

Figure 13 shows the angular distribution of the source energy in the forward direction.

Figure 14 shows the angular distribution of the source energy in the backward direction.

Figure 15 shows the angular distribution of the source energy in the forward direction.

at $x = 0$ and $y = 0$, and the contribution to the total energy is small. At low energy, the contribution to the total energy is small.

Thus, the contribution to the total energy is small.

Thus, the contribution to the total energy is small.

1. approximately isotropic
2. approximately independent of direction and energy in the backward direction
3. attenuated with the absorption coefficient of the source energy
4. independent of source energy.

Further, even at high energy, the contribution in the forward direction from scattering is considerable.

[illegible]

... are not met by airborne radiation detectors as they are usually oriented to minimize cosmic ray background from above and from the side. However, as the forward flux is little distorted by the presence of the detector, we can use the undisturbed γ flux at the detector as an apparent surface source density, i.e., consider Figure 6 which depicts a typical detector arrangement. We then treat the undisturbed forward flux at x , as arising from an active surface source $I(x, \lambda, z)$, in photons/unit area/

To see the difference between I_0 and the actual response of a shielded detector, consider a 100 percent efficient detector which receives all forward fluxes but none in the backward direction. This can be thought of as a thick crystal detector of large lateral extent so that edge effects are unimportant, and heavily shielded in the back. Then the count rate is simply $C(x, \lambda) = \int_0^\infty \lambda_0(x, \lambda) \exp(-\lambda_0 x) d\lambda_0$ and $I_0(x, \lambda, c)$ which we may call the current. In figure 1, we have computed the count rate, $C(x, \lambda)$, and the flux, $\lambda_0(x, \lambda)$, as a function of x due to a point C source emitting $\frac{1}{2\pi} \times 10^{10}$ / sec. / cm. of air and $\lambda_0 = 0.609$ mev; the flux is identical to equilibrium uranium flux, $\lambda_0 = 0.609$ mev for energies above 0.4 mev. Curves one and two are the primary and the current respectively due to the 0.609 mev gamma ray; as each curve is multiplied by a delta function, they are actually not comparable to the other curves in figure 1. These two curves may be thought of as representing the flux and current in the small interval λ_0 to $\lambda_0 + \delta\lambda$. In any event, it is evident that there is a large difference between the fluxes and the currents near the interface of source and air. With increasing distance from the source, the distance dependence of curves 1 and 2 decreases as much as some isotropy is attained. The sharp

drop in the fluxes (curves 3 and 5 in figure 7) near 1000 feet is not real as this drop is induced by the imposed boundary condition at 1000 feet.

The quantity measured by a radiation detector is dependent on the spectral energy and angular response of the detector, and is not the total flux but only a portion of the flux. Nevertheless, examination of the total flux characteristics should yield some important conclusions bearing on gamma measurements at considerable air distances from a source.

Figures 8, 9, and 10 display the total integral fluxes due to RaC, thorium, and K^{40} sources. The RaC source is normalized to $\frac{1}{4\pi}$ gamma rays/ft.³/sec., the thorium source to $\frac{1}{4\pi}$ disintegrations/ft.³/sec., and K^{40} source to $\frac{1}{4\pi}$ gamma rays/ft.³/sec. The dashed line curves denote the integral flux due solely to the unscattered photons. It is evident that the scattered flux forms the major part of the integral flux.

We can estimate the relative importance of Ra B to the integral flux from 2.4 mev. to 0.2 mev. and to the integral flux at 500 feet from the source E_0 to 0.2 mev., and to gamma rays of $\frac{1}{4\pi}$ gamma rays/ $\text{ft.}^2/\text{sec.}$ of energies 0.354, 0.297, and 0.243 were found to be .00063, .00053, and .00041 photons/ $\text{ft.}^2/\text{sec.}$ If we use the absolute intensity of the 2.2 mev. gamma ray estimated by Ellis and Aston (1930) and use Mladjencovic and Haggren's (1954) values of the relative intensities of the RaC gamma rays, we find that RaC emits about 1.42 primary gammas/disintegration. RaB emits 3 gamma rays of 0.354, 0.297, and 0.243 mev energies at the rate of 0.45, 0.258, and 0.115 gamma/disintegration (Ellis and Aston, 1930). Then the additional flux at 500 feet contributed by RaB is

$$\frac{(0.45 \times .00063) + (0.258 \times .00053) + (0.115 \times .00041)}{1.42} = 0.00033$$

photons/ $\text{ft.}^2/\text{sec.}$

or $\frac{.00033}{.007} \times 100 \sim 5$ percent additional flux. Thus, the contribution of RaB to the total flux is negligible.

...the value of λ_1 for which λ_1 is small and its
 ...is assumed, $\lambda_1 < 0.2$ mev. For the
 ...energy of 0.2 mev is emitted
 ...Let us assume we
 ...contribution to the
 integral flux in the range 0.2 mev $\pm$ 10 percent by this gamma ray is

$$\text{Primary flux} = \frac{1}{8\pi\lambda_1^2} E_0(\lambda_1) \lambda_1$$

$$\text{where } E_0(p) = \int_0^p \frac{e^{-pt}}{t^2} dt$$



the two order exponential integral. When the flux is evaluated at $x = 500$
 feet, and $\lambda_1 = 0.2$ mev ($\lambda_1 = 2.551$), it is found to be 1.2×10^{-4} photons/
 ft.²/sec. The contribution to the integral flux in the range 0.2 mev $\pm$ 10
 percent due to scattering from higher energy gamma rays is

$$\int_{0.18}^{0.22} N_0(x, E) dE = N_0(x, 0.18) - N_0(x, 0.22), \sim 0.92 \times 10^{-3} \text{ photons/ft.}^2/\text{sec.}$$

Taking into account the number of gamma rays per disintegration, we find that

$$\text{... yield } 100 \times \frac{1.2 \times 10^{-4}}{1.2 \times 10^{-3}} = 10 \text{ percent of the flux in the}$$

energy region 0.2 mev $\pm$ 10 percent. Considering the instability

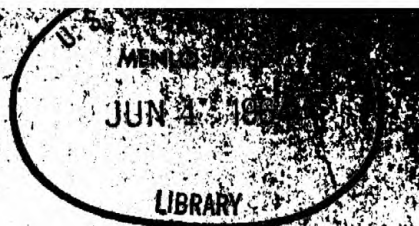
of the detector and statistical errors, it would be

difficult to detect such a line. It seems highly unlikely that spectral energy measurements at considerable air distances from a source could discriminate variations in equilibrium among the precursors of radium in the uranium series.

It is beyond the scope of this report to point out and to develop fully all possible applications of these particular solutions of the gamma-ray transport equation. A few possible uses are:

1. Establishment of the true total flux for various source concentrations, for the uranium, thorium, and potassium series, both individually and combined in varying amounts.
2. Comparison of the total flux, $Y(x,\lambda)$, to the observed, count-rate flux, $C(x,\lambda)$, for a particular technique of measurement.
3. Extension of the solutions to gamma-ray measurements at the ground surface and within the source (gamma-ray logging).
4. Evaluation of the possibility of spectral energy discrimination to determine the relative contributions from the uranium, thorium, and potassium series.

References cited.



1. Fano, U., 1953, Penetration of X- and gamma rays to extremely great depths: U. S. Nat. Bur. Standards Jour. Research, v. 51, no. 2, p. 95-121.
2. Goldstein, H., and Wilkins, J. E., 1954, Calculations of the penetration of gamma rays, final report: U. S. Atomic Energy Comm. NYO-3075, issued by U. S. Atomic Energy Comm. Tech. Inf. Service, Oak Ridge, Tenn.
3. Keller, H. B., and Heller, J., 1955, On the numerical integration of the neutron transport equation: U. S. Atomic Energy Comm. NYO-6481.
4. Mladjenevic, M., and Hedgran, A., 1954, An investigation of the
"Compton secondaries from RaC gamma rays: Arkiv fur Fysik, Band 8, Hefte 4, p. 49-63.
5. Ellis, C. D., and Aston, G. H., Proc. Rag. Soc. A129, 180, (1930).
6. Sakakura, A. Y., 1957, Scattered gamma rays from thick uranium sources, U. S. Geol. Survey Bull. 1052-A.

7. Certaine, J., 1953, Angular distribution of photons from plane

mono-energetic sources, U. S. Atomic Energy Comm. NYC-3074,

N.D.A. report 15C-10, contract AT(30-1)862.

~~Lazar,~~

8. ~~Lazar, N. H., private communication, 1955.~~

9. White, G. R., X-ray attenuation coefficients from 10 kev. to

100 mev.: N.B.S. report 1003, May 13, 1952.

10. Spencer, L. V., and Fano, U., J. Res. N.B.S. 46, 446 (1951).

Table: Coefficients of Legendre Moments

1
2
3
4

Accuracy of the
Legendre Moments

Moments	1st foot	2nd foot	3rd foot
ψ_0 ψ_0' $(\psi_0 - \psi_0')/\psi_0$	0.0003 0.0009 0.0006	0.0000 0.0000 0.0000	0.0018 0.0009 0.0009
ψ_1 ψ_1' $(\psi_1 - \psi_1')/\psi_1$	0.0006 0.0005 0.0003	0.0000 0.0000 0.0000	0.0013 0.0006 0.0006
ψ_2 ψ_2' $(\psi_2 - \psi_2')/\psi_2$	0.0 0.0005 <u>0.0005</u>	0.0003 0.0005 0.0002	0.0006 0.0006 0.0000
ψ_3 ψ_3' $(\psi_3 - \psi_3')/\psi_3$	-0.0003 -0.0003 0.0000	-0.0003 -0.0006 0.0000	0.0006 0.0006 0.0000
ψ_4 ψ_4' $(\psi_4 - \psi_4')/\psi_4$	0.0 0.0003 <u>0.0003</u>	-0.0006 -0.0007 0.0000	-0.0006 0.0006 0.0000

7

62

	1	2	3	4	5	6	7
1	1.196	1.126	1.136	1.0911	1.076	0.1066	
2	1.196	1.111	1.118	1.117	1.118	0.1066	
3	1.196	1.0987	1.107	1.116	1.108	0.1066	
4	1.196	1.0858	1.0809	1.088	1.080	0.1066	
5	1.196	1.0728	1.0685	1.07196	1.0682	0.1066	
6	1.196	1.0599	1.0559	1.0576	1.0582	0.1066	
7	1.196	1.0469	1.0438	1.04578	1.0468	0	
8	1.196	1.0339	1.0315	1.03373	1.0340	0	
9	1.196	1.0209	1.0196	1.02136	1.0218	0	
10	1.196	1.0079	1.0074	1.00904	1.0093	0	
11	1.196	0.9949	0.9951	0.9967	0.997	0	
12	1.196	0.9819	0.983	0.98463	0.985	0	
13	1.196	0.9689	0.971	0.97266	0.973	0	
14	1.196	0.9559	0.959	0.9576	0.9582	0	
15	1.196	0.9429	0.947	0.94578	0.9468	0	
16	1.196	0.9299	0.935	0.93373	0.9340	0	
17	1.196	0.9169	0.923	0.92136	0.9218	0	
18	1.196	0.9039	0.911	0.90904	0.9093	0	
19	1.196	0.8909	0.900	0.8967	0.897	0	
20	1.196	0.8779	0.887	0.88463	0.885	0	
21	1.196	0.8649	0.875	0.87266	0.873	0	
22	1.196	0.8519	0.863	0.86063	0.861	0	
23	1.196	0.8389	0.851	0.84863	0.849	0	
24	1.196	0.8259	0.839	0.83373	0.8340	0	
25	1.196	0.8129	0.827	0.8173	0.818	0	
26	1.196	0.8000	0.815	0.80463	0.805	0	
27	1.196	0.7870	0.803	0.79266	0.793	0	
28	1.196	0.7740	0.791	0.78063	0.781	0	
29	1.196	0.7610	0.779	0.76863	0.769	0	
30	1.196	0.7480	0.767	0.75663	0.757	0	
31	1.196	0.7350	0.755	0.74463	0.745	0	
32	1.196	0.7220	0.743	0.73263	0.733	0	
33	1.196	0.7090	0.731	0.72063	0.721	0	
34	1.196	0.6960	0.719	0.70863	0.709	0	
35	1.196	0.6830	0.707	0.69663	0.697	0	
36	1.196	0.6700	0.695	0.68463	0.685	0	
37	1.196	0.6570	0.683	0.67266	0.673	0	
38	1.196	0.6440	0.671	0.66063	0.661	0	
39	1.196	0.6310	0.659	0.64863	0.649	0	
40	1.196	0.6180	0.647	0.63663	0.637	0	
41	1.196	0.6050	0.635	0.62463	0.625	0	
42	1.196	0.5920	0.623	0.61263	0.613	0	
43	1.196	0.5790	0.611	0.60063	0.601	0	
44	1.196	0.5660	0.599	0.58863	0.589	0	
45	1.196	0.5530	0.587	0.57663	0.577	0	
46	1.196	0.5400	0.575	0.56463	0.565	0	
47	1.196	0.5270	0.563	0.55263	0.553	0	
48	1.196	0.5140	0.551	0.54063	0.541	0	
49	1.196	0.5010	0.539	0.52863	0.529	0	
50	1.196	0.4880	0.527	0.51663	0.517	0	

[illegible]

1	2	3	4	5	6	7	8
1	2	3	4	5	6	7	8
9	10	11	12	13	14	15	16
17	18	19	20	21	22	23	24
25	26	27	28	29	30	31	32
33	34	35	36	37	38	39	40
41	42	43	44	45	46	47	48
49	50	51	52	53	54	55	56
57	58	59	60	61	62	63	64
65	66	67	68	69	70	71	72
73	74	75	76	77	78	79	80
81	82	83	84	85	86	87	88
89	90	91	92	93	94	95	96
97	98	99	100	101	102	103	104
105	106	107	108	109	110	111	112
113	114	115	116	117	118	119	120
121	122	123	124	125	126	127	128
129	130	131	132	133	134	135	136
137	138	139	140	141	142	143	144
145	146	147	148	149	150	151	152
153	154	155	156	157	158	159	160
161	162	163	164	165	166	167	168
169	170	171	172	173	174	175	176
177	178	179	180	181	182	183	184
185	186	187	188	189	190	191	192
193	194	195	196	197	198	199	200
201	202	203	204	205	206	207	208
209	210	211	212	213	214	215	216
217	218	219	220	221	222	223	224
225	226	227	228	229	230	231	232
233	234	235	236	237	238	239	240
241	242	243	244	245	246	247	248
249	250	251	252	253	254	255	256
257	258	259	260	261	262	263	264
265	266	267	268	269	270	271	272
273	274	275	276	277	278	279	280
281	282	283	284	285	286	287	288
289	290	291	292	293	294	295	296
297	298	299	300	301	302	303	304
305	306	307	308	309	310	311	312
313	314	315	316	317	318	319	320
321	322	323	324	325	326	327	328
329	330	331	332	333	334	335	336
337	338	339	340	341	342	343	344
345	346	347	348	349	350	351	352
353	354	355	356	357	358	359	360
361	362	363	364	365	366	367	368
369	370	371	372	373	374	375	376
377	378	379	380	381	382	383	384
385	386	387	388	389	390	391	392
393	394	395	396	397	398	399	400
401	402	403	404	405	406	407	408
409	410	411	412	413	414	415	416
417	418	419	420	421	422	423	424
425	426	427	428	429	430	431	432
433	434	435	436	437	438	439	440
441	442	443	444	445	446	447	448
449	450	451	452	453	454	455	456
457	458	459	460	461	462	463	464
465	466	467	468	469	470	471	472
473	474	475	476	477	478	479	480
481	482	483	484	485	486	487	488
489	490	491	492	493	494	495	496
497	498	499	500	501	502	503	504
505	506	507	508	509	510	511	512
513	514	515	516	517	518	519	520
521	522	523	524	525	526	527	528
529	530	531	532	533	534	535	536
537	538	539	540	541	542	543	544
545	546	547	548	549	550	551	552
553	554	555	556	557	558	559	560
561	562	563	564	565	566	567	568
569	570	571	572	573	574	575	576
577	578	579	580	581	582	583	584
585	586	587	588	589	590	591	592
593	594	595	596	597	598	599	600
601	602	603	604	605	606	607	608
609	610	611	612	613	614	615	616
617	618	619	620	621	622	623	624
625	626	627	628	629	630	631	632
633	634	635	636	637	638	639	640
641	642	643	644	645	646	647	648
649	650	651	652	653	654	655	656
657	658	659	660	661	662	663	664
665	666	667	668	669	670	671	672
673	674	675	676	677	678	679	680
681	682	683	684	685	686	687	688
689	690	691	692	693	694	695	696
697	698	699	700	701	702	703	704
705	706	707	708	709	710	711	712
713	714	715	716	717	718	719	720
721	722	723	724	725	726	727	728
729	730	731	732	733	734	735	736
737	738	739	740	741	742	743	744
745	746	747	748	749	750	751	752
753	754	755	756	757	758	759	760
761	762	763	764	765	766	767	768
769	770	771	772	773	774	775	776
777	778	779	780	781	782	783	784
785	786	787	788	789	790	791	792
793	794	795	796	797	798	799	800
801	802	803	804	805	806	807	808
809	810	811	812	813	814	815	816
817	818	819	820	821	822	823	824
825	826	827	828	829	830	831	832
833	834	835	836	837	838	839	840
841	842	843	844	845	846	847	848
849	850	851	852	853	854	855	856
857	858	859	860	861	862	863	864
865	866	867	868	869	870	871	872
873	874	875	876	877	878	879	880
881	882	883	884	885	886	887	888
889	890	891	892	893	894	895	896
897	898	899	900	901	902	903	904
905	906	907	908	909	910	911	912
913	914	915	916	917	918	919	920
921	922	923	924	925	926	927	928
929	930	931	932	933	934	935	936
937	938	939	940	941	942	943	944
945	946	947	948	949	950	951	952
953	954	955	956	957	958	959	960
961	962	963	964	965	966	967	968
969	970	971	972	973	974	975	976
977	978	979	980	981	982	983	984
985	986	987	988	989	990	991	992
993	994	995	996	997	998	999	1000

Table 2 (contd)

$\frac{x}{L} = 0.0$		0	1	2	3	4	Grid
1		.009108	.01918	.01968	.01995	.02778	.01861
1/2		.009108	.01735	.01760	.01764	.02356	.01718
1/3		.009108	.01598	.01557	.01549	.01890	.01597
2/3		.009108	.01369	.01359	.01347	.01298	.01396
1/2		.009108	.01186	.01166	.01157	.01185	.01295
1/2		.009108	.01008	.009779	.009763	.01055	.01036
-1/2		.009108	.008197	.007967	.007932	.008756	.007860
-2/3		.009108	.006360	.006165	.006259	.006520	.005826
-1/2		.009108	.004598	.004633	.004551	.004056	.004150
-1/2		.009108	.002896	.002790	.002932	.001838	.002660
-1/2		.009108	.001118	.001118	.001080	.0006722	.001171
-1		.009108	.000676	.000651	.000724	.001606	.000591

$\frac{x}{L} = 0.01$		0	1	2	3	4	Grid
1		.003886	.009703	.01260	.01133	.01657	.01382
1/2		.003886	.00865	.01200	.01088	.009985	.0101
1/3		.003886	.007487	.007876	.007276	.006321	.007100
2/3		.003886	.006499	.006016	.005192	.004670	.005282
1/2		.003886	.005671	.004676	.003733	.004003	.004669
1/2		.003886	.004123	.003098	.002939	.003612	.003907
-1/2		.003886	.003395	.002060	.002300	.003073	.002757
-2/3		.003886	.002897	.001850	.002461	.002711	.001866
-1/2		.003886	.002239	.0007265	.002588	.001110	.001139
-1/2		.003886	.001810	.0006707	.002069	.0007121	.0005299
-1/2		.003886	.0008769	.0006822	.0008066	.0001862	.0001789
-1		.003886	.001935	.0007611	.001167	.000779	.00026673

$\frac{x}{L} = 0.001$		0	1	2	3	4	Grid
1		.001708	.009017	.007065	.007637	.008535	.008078
1/2		.001708	.008615	.008668	.005529	.005372	.005379
1/3		.001708	.008316	.008036	.003856	.003676	.003695
2/3		.001708	.008018	.007823	.002967	.002376	.002753
1/2		.001708	.007621	.007515	.001610	.001718	.002883
1/2		.001708	.007209	.007020	.0009331	.001613	.001801
-1/2		.001708	.006807	.006606	.0006895	.0007668	0
-1/2		.001708	.0064059	.0062071	.0002151	.0003838	0
-2/3		.001708	.0060063	.0058090	.00007055	.0001208	0
-1/2		.001708	.0056072	.00540771	.00001883	.00003875	0
-1/2		.001708	.0052087	.005003971	.000006728	.00000643	0
-1		.001708	.0048000	.00460681	.00001236	.00007766	0

Table 2 (cont'd)

$L = 1$		0	1	2	3	4	Ar101
I=0	1	.006878	.007352	.007952	.007903	.008621	.007619
	1/2	.006878	.007364	.007367	.007617	.007216	.007887
	1/3	.006878	.007288	.007888	.007098	.006360	.007288
	1/4	.006878	.007096	.007096	.006899	.006533	.007008
	1/5	.006878	.007009	.006930	.006806	.007021	.006860
	1/6	.006878	.006972	.006826	.006777	.007368	.006638
	1/7	.006878	.006835	.006738	.006755	.007377	.007886
	1/8	.006878	.006769	.006672	.006798	.007006	.006963
	1/9	.006878	.006663	.006685	.006782	.006616	.006767
	1/10	.006878	.006577	.006598	.006707	.005975	.006859
	1/11	.006878	.006492	.006591	.006642	.006761	.006393
	1	.006878	.006405	.006406	.006753	.007972	.006822

$L = 1000$		0	1	2	3	4	Ar101
I=500	1	.002788	.003137	.003890	.004078	.004767	.003718
	1/2	.002788	.003173	.003653	.003688	.003363	.003608
	1/3	.002788	.003020	.003092	.003037	.002768	.003093
	1/4	.002788	.002964	.002803	.002729	.002577	.002886
	1/5	.002788	.002883	.002758	.002523	.002603	.002978
	1/6	.002788	.002819	.002657	.002627	.002657	.002668
	1/7	.002788	.002756	.002689	.002611	.002611	.002689
	1/8	.002788	.002699	.002608	.002667	.002768	.002668
	1/9	.002788	.002639	.002685	.002770	.002637	.002638
	1/10	.002788	.002545	.002666	.002706	.002619	.002717
	1/11	.002788	.002482	.002681	.002696	.002738	.002603
	1	.002788	.002438	.002692	.003003	.003678	.002817

$L = 10000$		0	1	2	3	4	Ar101
I=2000	1	.000618	.000670	.000999	.001775	.002933	.001720
	1/2	.000618	.000679	.000665	.000613	.000595	.000695
	1/3	.000618	.000688	.000673	.000693	.000625	.000616
	1/4	.000618	.000696	.000636	.000616	.000602	.000685
	1/5	.000618	.000697	.0006773	.0006979	.000670	.000689
	1/6	.000618	.000710	.000675	.000637	.0006383	.000676
	1/7	.000618	.0005378	.0006378	.0006380	.0006867	0
	1/8	.000618	.000636	.0006037	.0006131	.0006179	0
	1/9	.000618	.0006106	.00067778	.0006768	.0006669	0
	1/10	.000618	.0006083	.0006145	.00069717	.0006528	0
	1/11	.000618	.0006120	.00067608	.0006611	.00067186	0
	1	.000618	.0006333	.0006039	.0006708	.0006795	0

File 2 (cont'd)

	0	1	2	3	4	AP101
1	.06029	.007079	.07227	.007266	.004782	.006933
9/11	.06029	.006888	.004963	.006968	.006703	.006806
7/11	.004029	.006497	.006713	.06701	.004055	.006479
5/11	.006029	.006906	.006678	.006661	.006138	.006466
3/11	.006029	.06115	.006798	.006216	.00426	.006388
1/11	.006029	.06126	.006052	.004067	.06569	.006803
-1/11	.006029	.005933	.005861	.0547	.06389	.006080
-3/11	.06029	.005762	.005685	.005699	.005882	.005736
-5/11	.006029	.005551	.005523	.005461	.005218	.005510
-7/11	.006029	.005360	.005376	.005388	.004742	.005322
-9/11	.006029	.005160	.005266	.005238	.004973	.005158
-1	.006029	.004979	.005126	.004957	.006403	.004988

	0	1	2	3	4	AP101
1	.003666	.003966	.003775	.04005	.004671	.003649
9/11	.003666	.003152	.003009	.003167	.003386	.003188
7/11	.003666	.003039	.003096	.003073	.003739	.003077
5/11	.003666	.002927	.002930	.002777	.002985	.002980
3/11	.003666	.002815	.002766	.002936	.002616	.002972
1/11	.003666	.002702	.002653	.002622	.002651	.002667
-1/11	.003666	.002590	.002661	.002771	.002602	.002388
-3/11	.003666	.002479	.002779	.002661	.002611	.002366
-5/11	.003666	.002365	.002668	.002370	.002779	.002387
-7/11	.003666	.002253	.002305	.002379	.002095	.002386
-9/11	.003666	.002141	.002398	.002366	.002769	.002317
-1	.003666	.002028	.002280	.002310	.002977	.002260

	0	1	2	3	4	AP101
1	.0006178	.001668	.001995	.0174	.001925	0.028713
9/11	.0006178	.001677	.001668	.001610	.001983	.001528
7/11	.0006178	.001296	.001371	.001398	.001325	.001335
5/11	.0006178	.001095	.001033	.001136	.001101	.001185
3/11	.0006178	.000906	.000777	.000956	.000977	.001029
1/11	.0006178	.000713	.000537	.000962	.0006381	.0009673
-1/11	.0006178	.000523	.000627	.000337	.0003461	0
-3/11	.0006178	.000336	.0007063	.0001230	.0003615	0
-5/11	.0006178	.0001405	.00007927	.00002317	.0000690	0
-7/11	.0006178	.0000406	.00001528	.0000468	.0000526	0
-9/11	.0006178	.00003616	.00007637	.0000415	.00007157	0
-1	.0006178	.0000323	.00001090	.00001719	.0000786	0

Table 2 (cont'd)

L		0	1	2	3	4	trial
2	1	.005333	.007311	.007152	.006166	.011661	.006162
	9/11	.005333	.007176	.006206	.006208	.003171	.006066
	7/11	.005333	.006950	.005973	.005968	.0002701	.005956
	5/11	.005333	.005779	.004753	.004717	.000176	.005124
	3/11	.005333	.004401	.003517	.003512	.007111	.005665
	1/11	.005333	.003123	.002355	.002353	.000457	.004476
	-1/11	.005333	.002214	.00176	.00178	.000783	.005315
	-3/11	.005333	.005004	.003022	.005017	.000226	.005053
	-5/11	.005333	.003001	.001141	.001157	.000019	.001114
	-7/11	.005333	.001708	.001724	.001729	-.000946	.001665
	-9/11	.005333	.001531	.001601	.001509	.000262	.001522
	-1	.005333	.001352	.001198	.001177	.00115	.001397

L		0	1	2	3	4	trial
2	1	.002312	.003011	.003311	.003111	.001113	.003217
	9/11	.002312	.002981	.003011	.003113	.003003	.003035
	7/11	.002312	.002602	.002536	.002706	.002530	.002820
	5/11	.002312	.002612	.002617	.002559	.002124	.002618
	3/11	.002312	.002562	.002629	.002302	.002415	.002116
	1/11	.002312	.002112	.002275	.002708	.002471	.002312
	-1/11	.002312	.003322	.002155	.002173	.002309	
	-3/11	.002312	.002702	.000669	.002116	.002191	
	-5/11	.002312	.002112	.000017	.002075	.001911	
	-7/11	.002312	.001962	.001999	.002039	.002771	
	-9/11	.002312	.001812	.000015	.001996	.001166	
	-1	.002312	.001722	.000065	.001935	.002561	

L		0	1	2	3	4	trial
2	1	.0005106	.001164	.001760	.001565	.001695	.001511
	9/11	.0005106	.001196	.001115	.001116	.001395	.001337
	7/11	.0005106	.001188	.001160	.001221	.001165	.001169
	5/11	.0005106	.0009601	.0009011	.000912	.000937	.000939
	3/11	.0005106	.0007985	.0006771	.0007173	.0007628	.000804
	1/11	.0005106	.0006214	.0001601	.0005061	.0005509	.000736
	-1/11	.0005106	.0001567	.0003123	.0002660	.0003306	.0
	-3/11	.0005106	.0002008	.0001738	.0001079	.0001146	.0
	-5/11	.0005106	.0001209	.0000462	.0000225	.0000008	.0
	-7/11	.0005106	-.00001645	-.0000012	-.0000711	.0001307	.0
	-9/11	.0005106	-.0000118	-.0000010	.0000011	.0000011	.0
	-1	.0005106	-.0000127	-.0000011	.0000005	.0000008	.0

Table 3. Comparison of Ariel and Moments methods.

<u>X =</u>	<u>Moments</u>	<u>Ariel</u>
0'	2.08	2.06
100'	2.41	2.31
300'	2.87	2.74
500'	3.51	3.11
700'	3.63	3.47

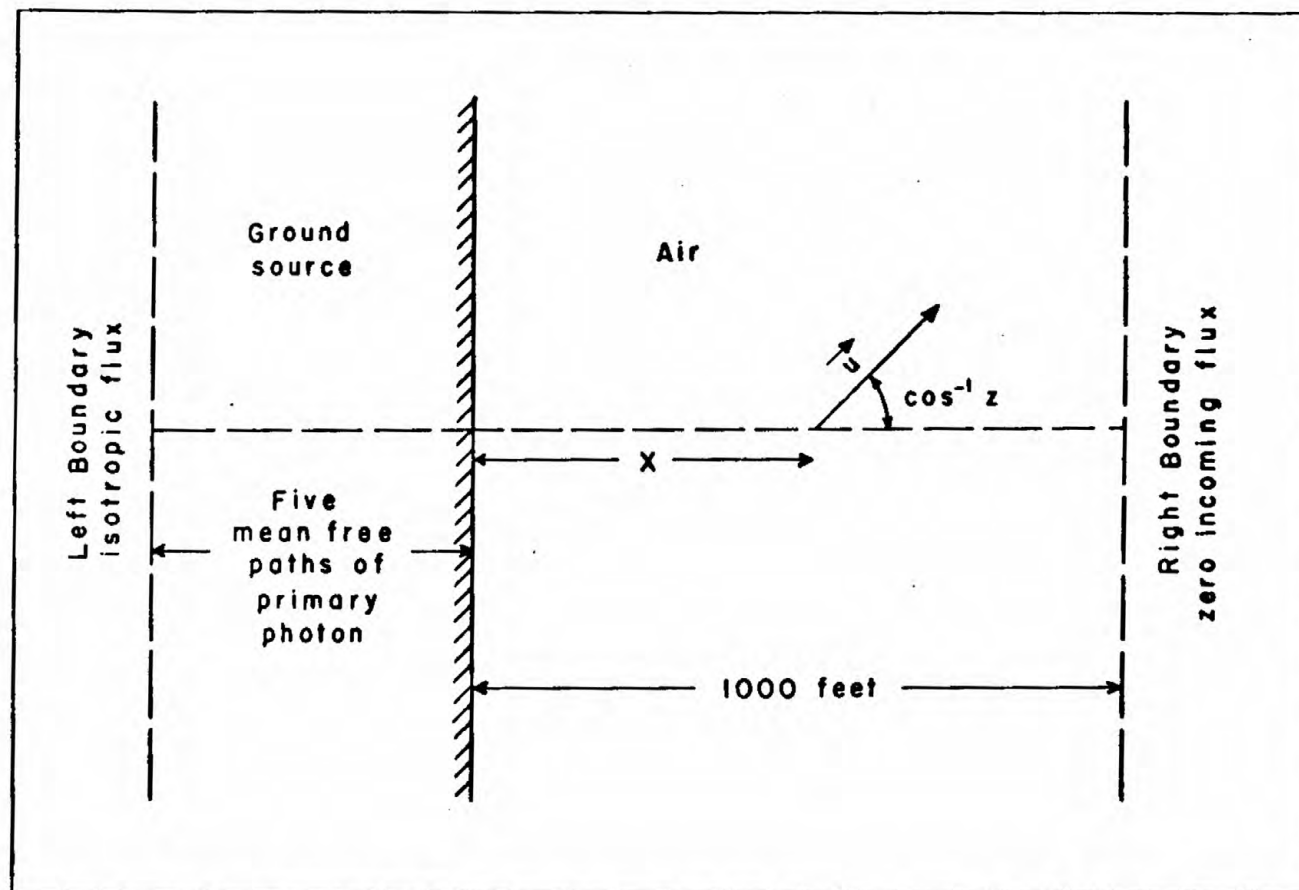
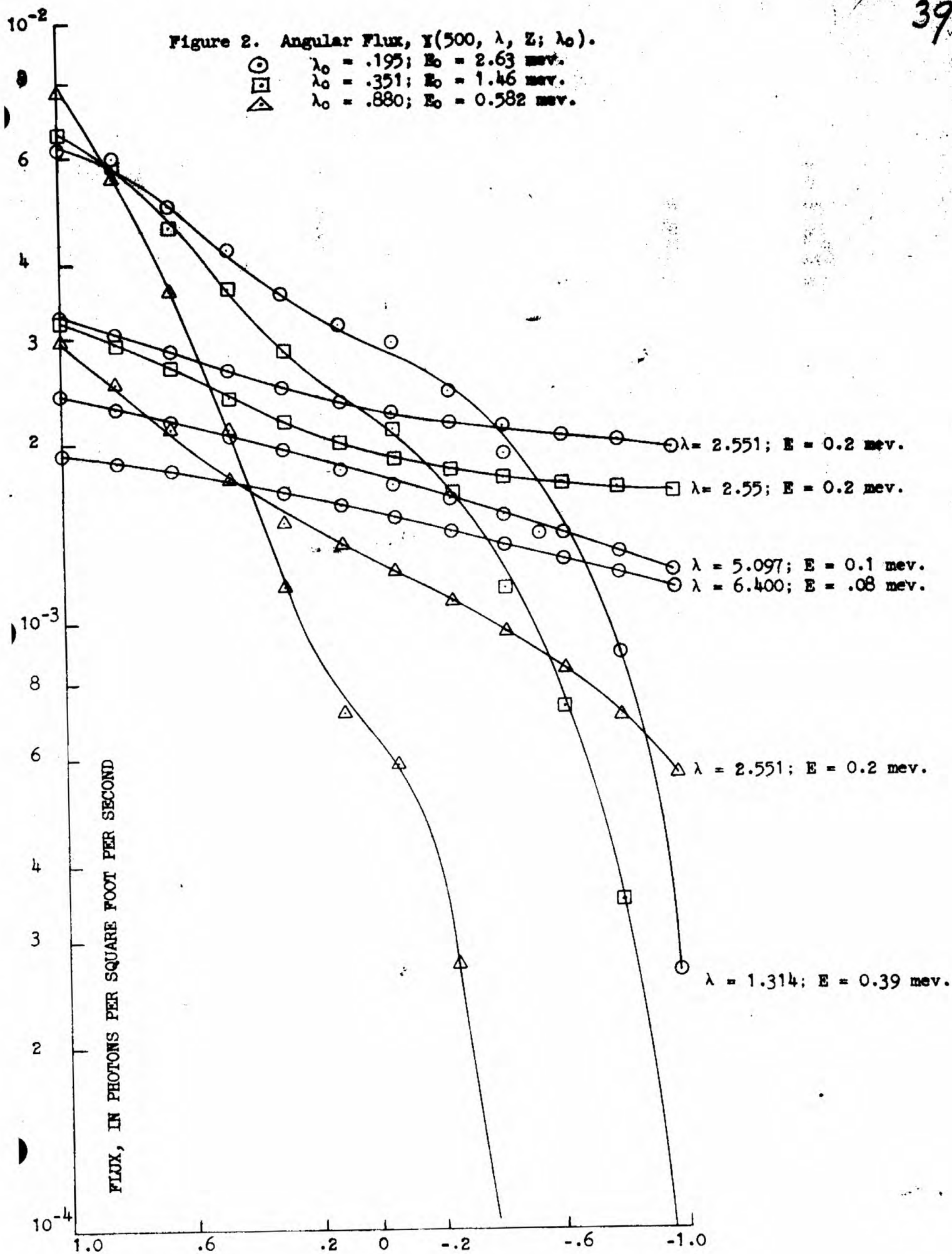


Figure 1- Geometry for two media, ground and air

Figure 2. Angular Flux, $\Upsilon(500, \lambda, Z; \lambda_0)$.

$\odot$ $\lambda_0 = .195; E_0 = 2.63$ mev.
 $\square$ $\lambda_0 = .351; E_0 = 1.46$ mev.
 $\triangle$ $\lambda_0 = .880; E_0 = 0.582$ mev.



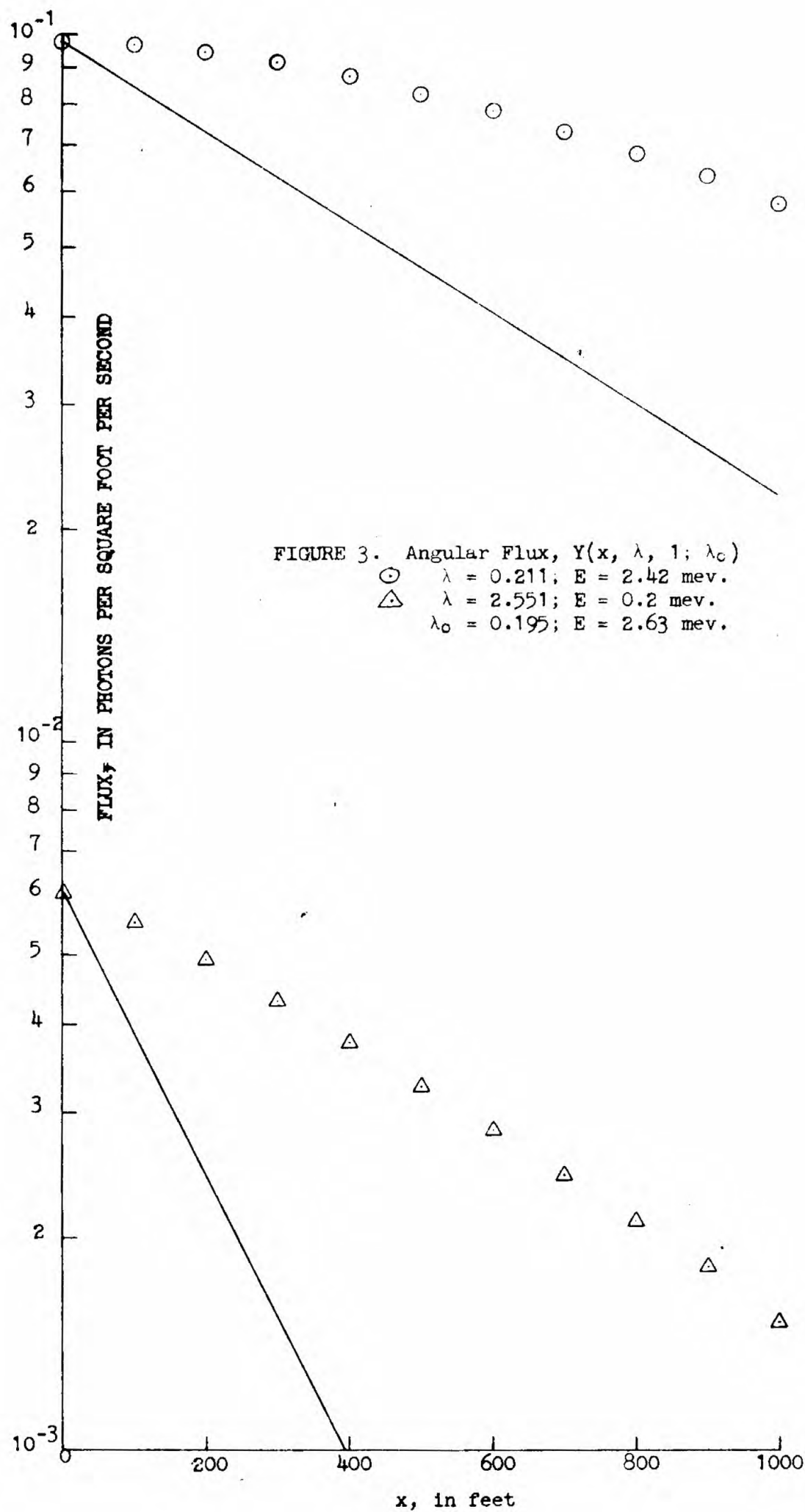
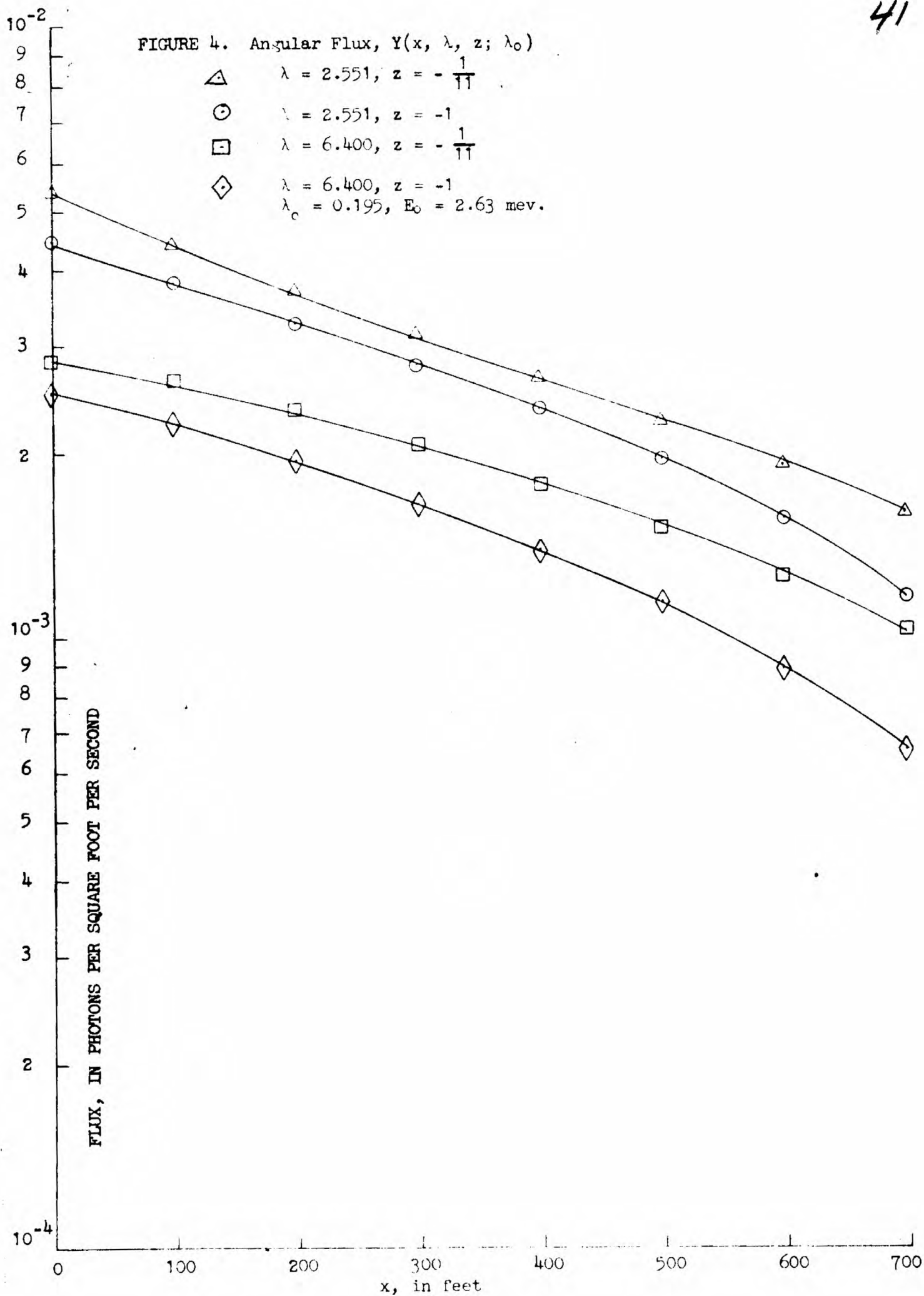
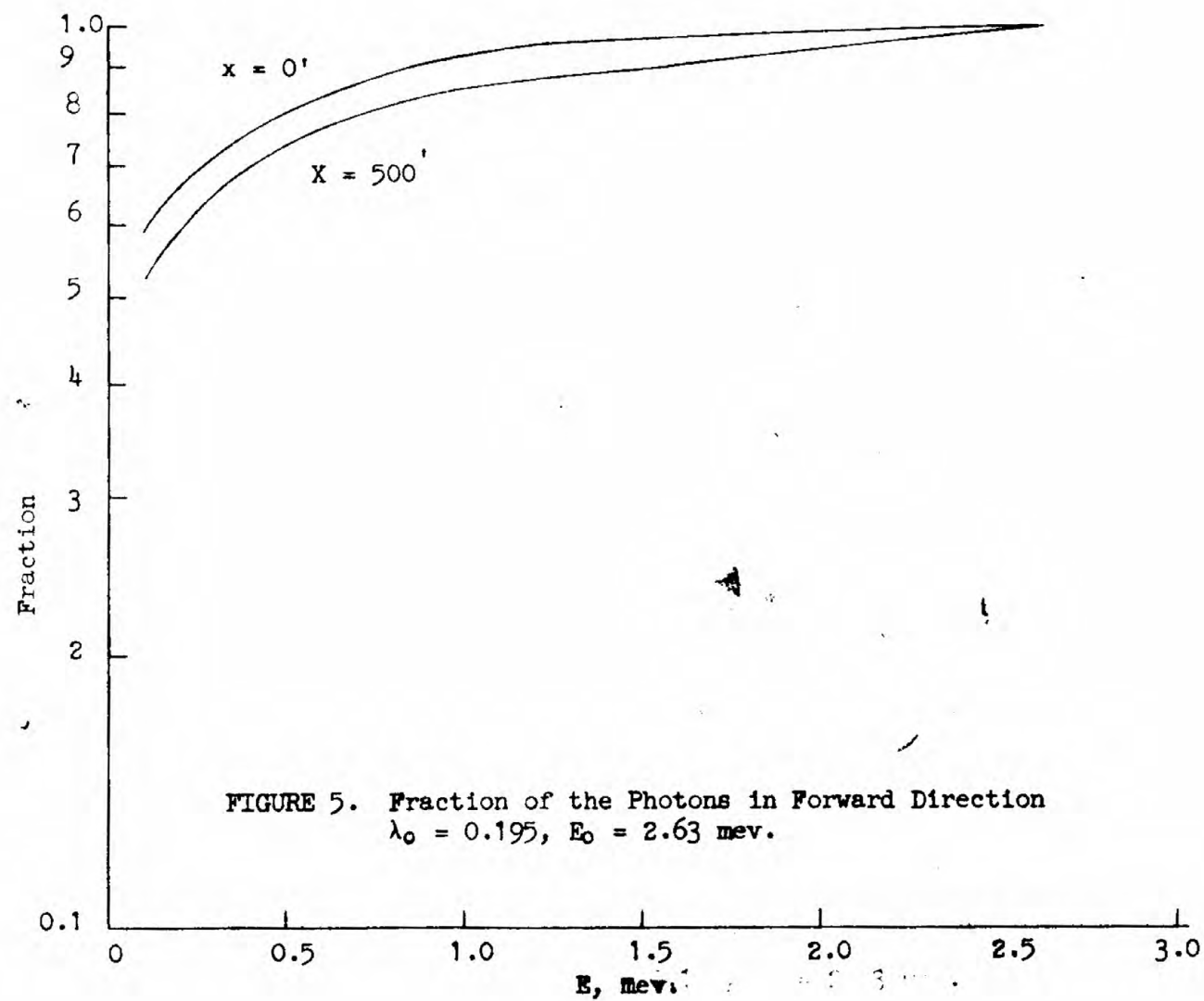


FIGURE 4. Angular Flux, $Y(x, \lambda, z; \lambda_0)$ 



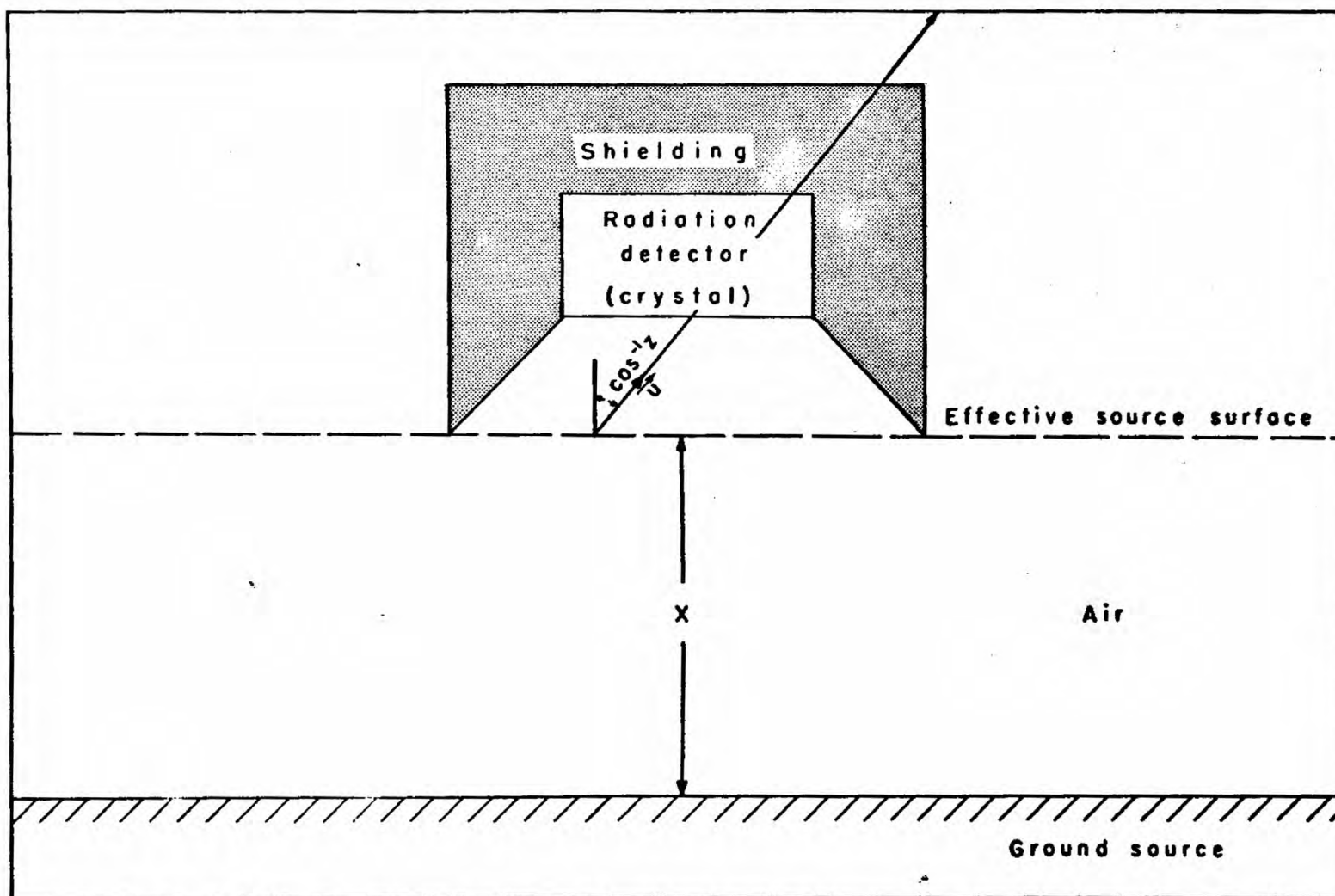


Figure 6 Effective source surface for airborne radiation detector

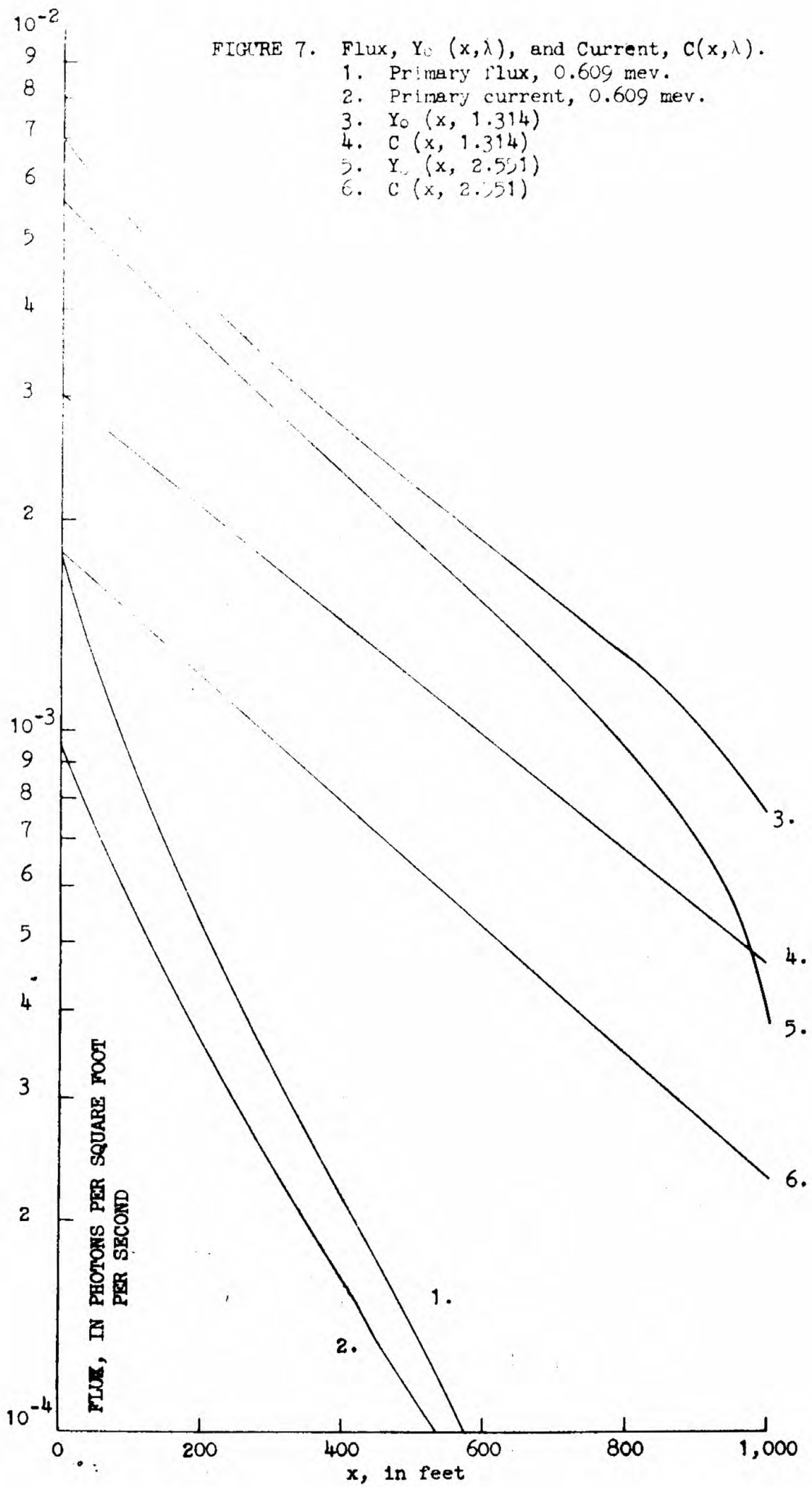


FIGURE 8. Integral Fluxes at 500 feet due to RaC
 — Total integral flux, $N_C(500, E)$
 - - - Primary integral flux

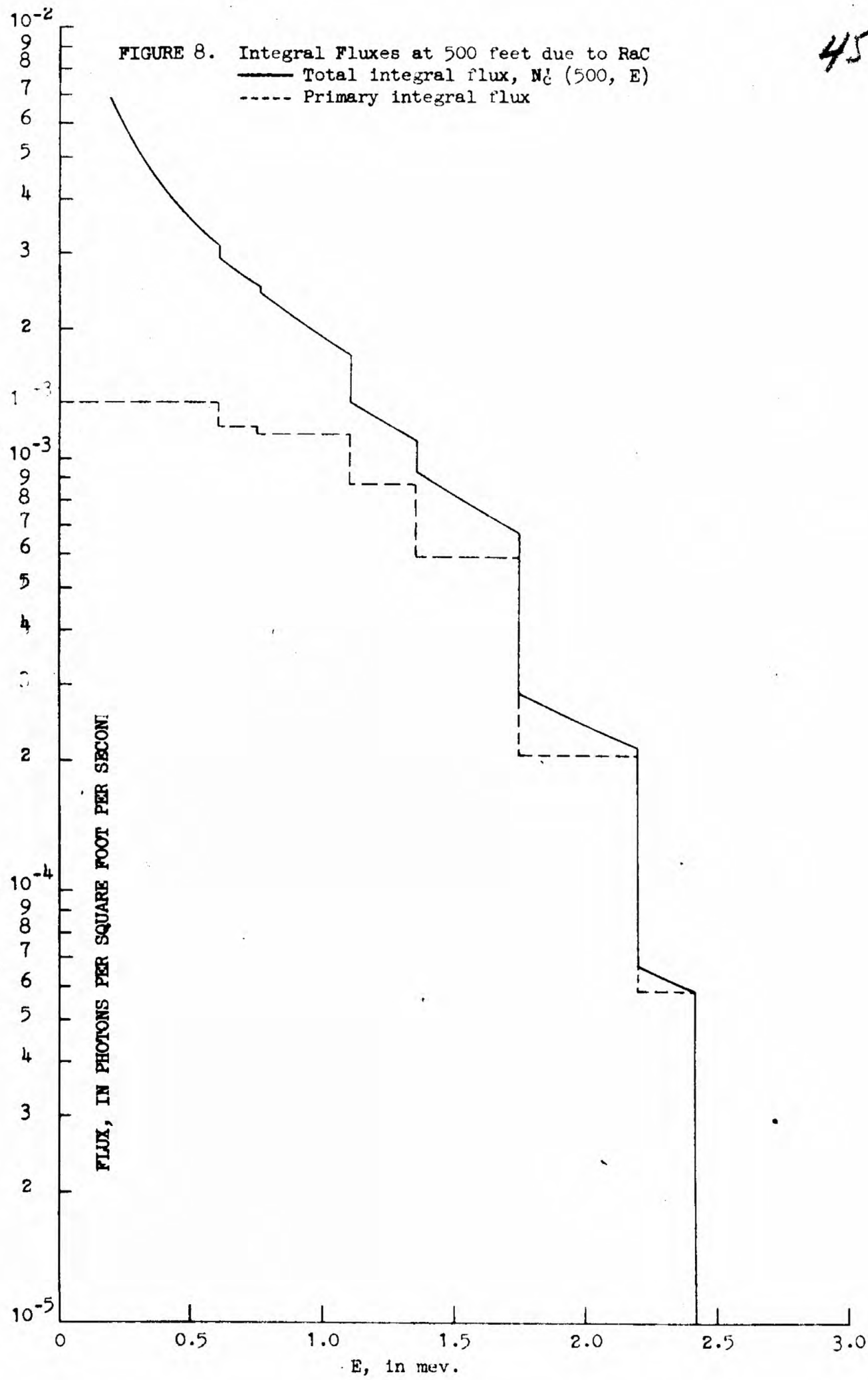
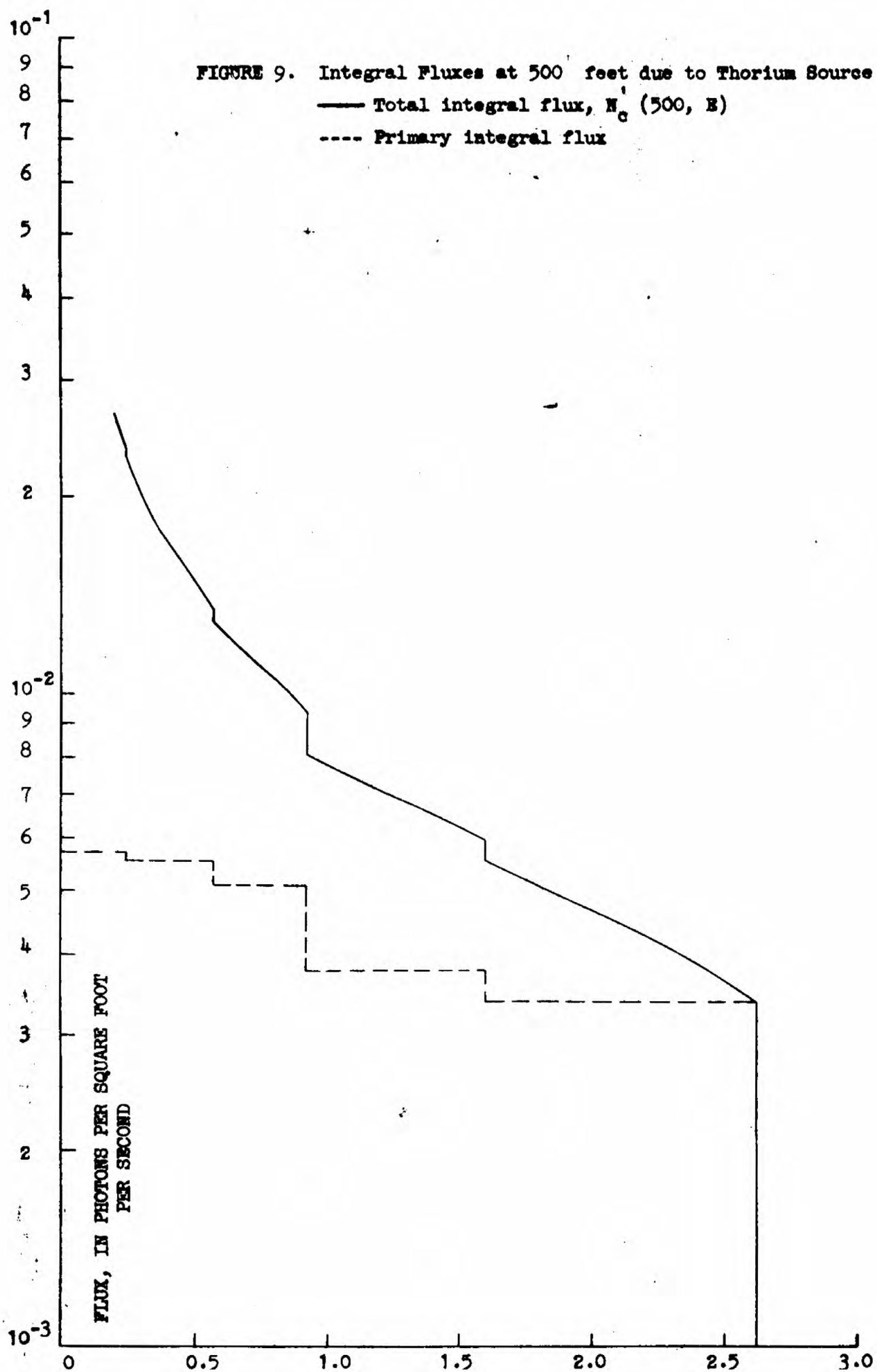
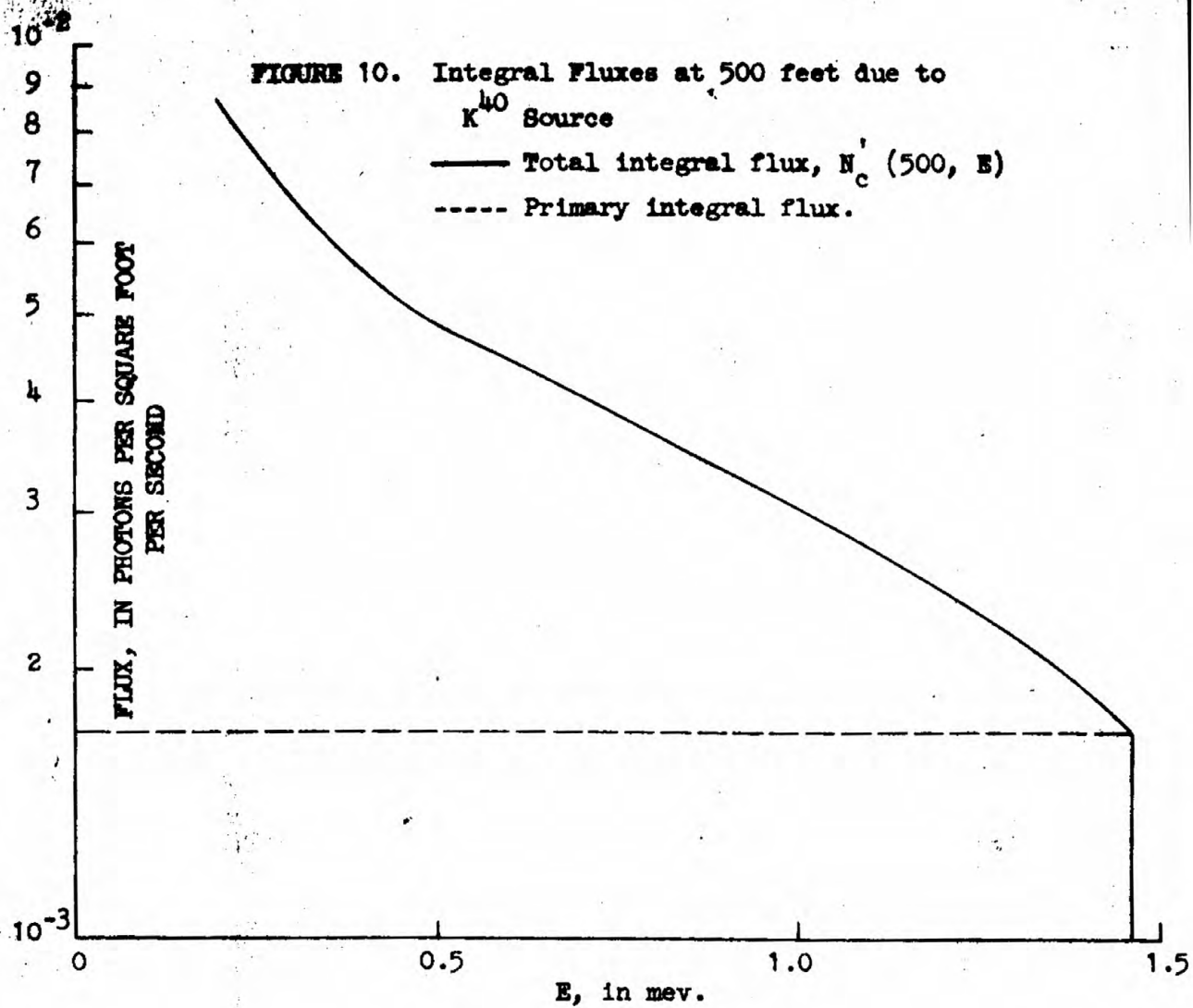


FIGURE 9. Integral Fluxes at 500 feet due to Thorium Source





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